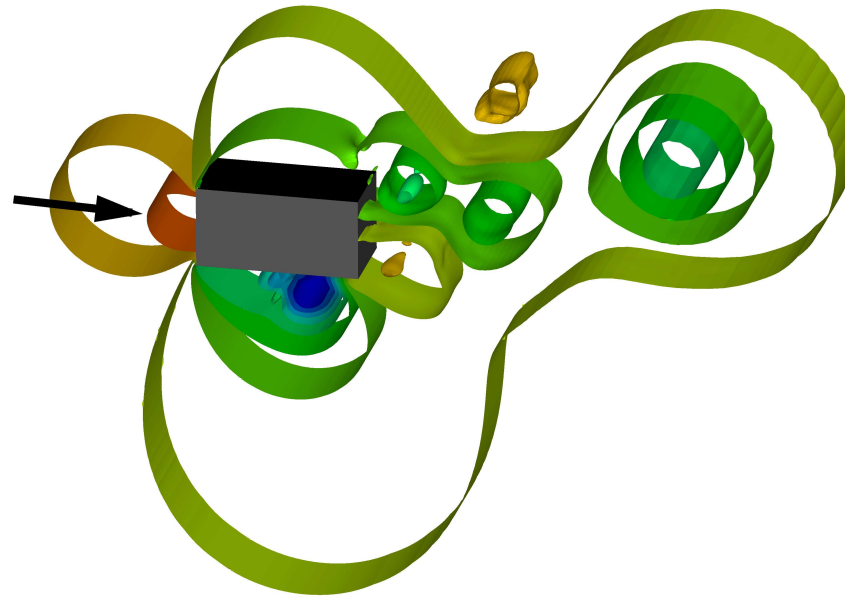


Simulation strömungsinduzierter Schwingungen eines rechteckigen Körpers

U. Bunge, A. Gurr, F. Thiele

TU Berlin, Germany



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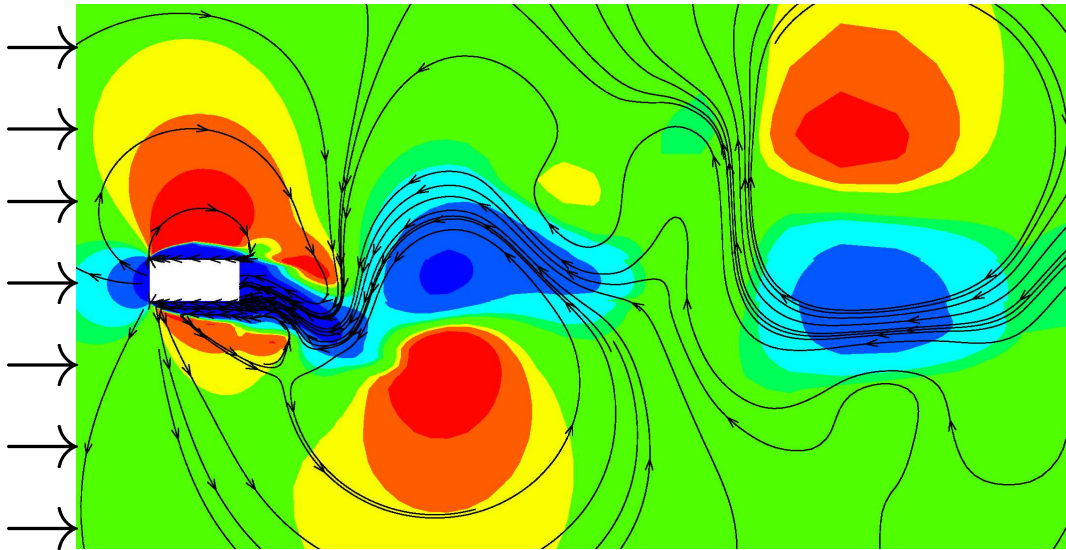
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HFI SEMINAR
FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY



- **Einleitung und Ziel der Simulationen**
 - Beschreibung der physikalischen Vorgänge
 - Berechnung der Resonanz–Strömungsgeschwindigkeit mit validierten Methoden
 - Identifikation der (numerischen) Einflußparameter
 - Bewertung des URANS–Ansatzes (Experimentelle Daten)
- **Wie wurde das erreicht? (Methode)**
 - Simple Model, 2D/3D, URANS, DES (standard), explizite Kopplung
- **Was wurde bis jetzt erreicht? (Ergebnisse)**
 - Unbewegt 2D/3D, URANS, DES
 - Schwingend 2D, URANS
 - Vergleich mit Experimenten
- **Zusammenfassung, Schlussfolgerungen und Ausblick**
- **Erste neue Ergebnisse, DES (Zwei–Gleichungs–Modell)**





- Geometry of bluff body

$$L/H = 2$$

- High Reynolds number

$$Re = UH/\nu = 1 \cdot 10^4 \dots 6 \cdot 10^4$$

⇒ Vortex street

Two dominant frequencies:

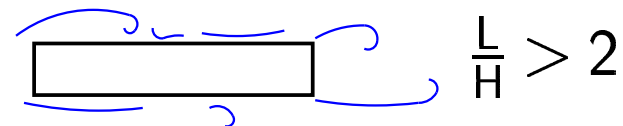
- Vortex shedding, f_s ($\frac{L}{H} < 2$)

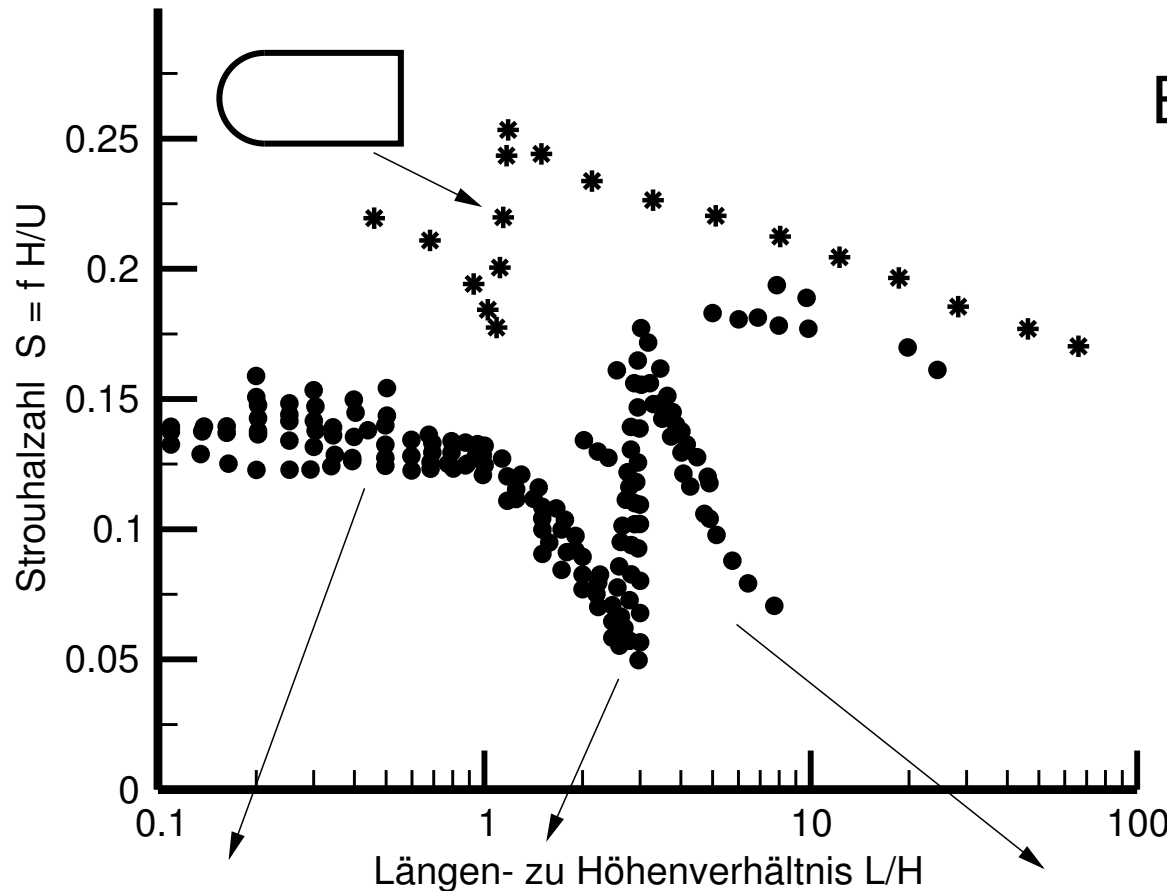
$$St_s = f_s H / U$$

- Vortex impingement, $f_i > f_s$

$$St_i = f_i H / U > St_s$$

can cause resonance of free body



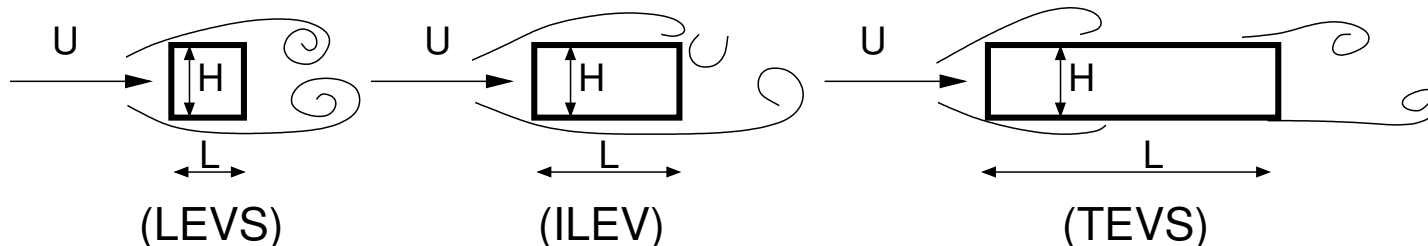


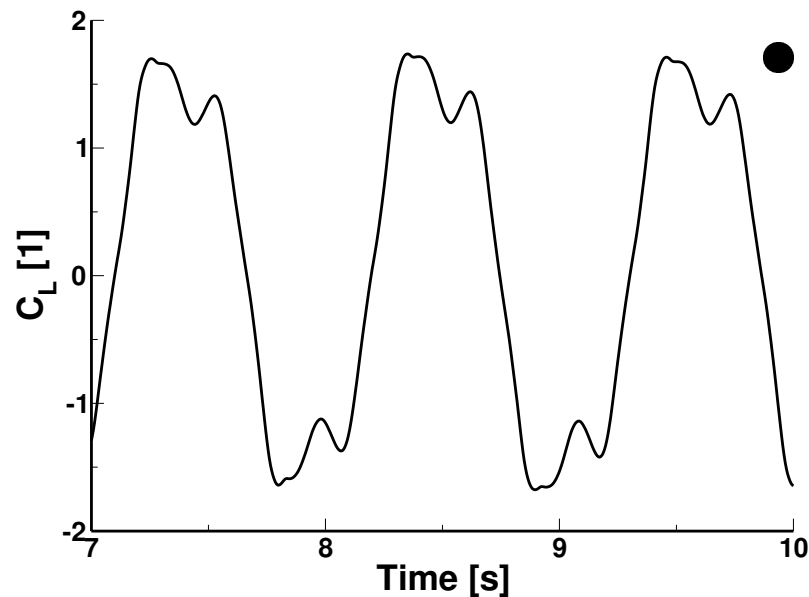
Experimentelle Daten

- Unbewegte Körper
- Re-Zahlen $O(10^4 - 10^5)$
- LEVS und TEVS
- Indifferenten Zustand bei $L/H \approx 2 - L/H \approx 5$

Sprünge in S

ILEV

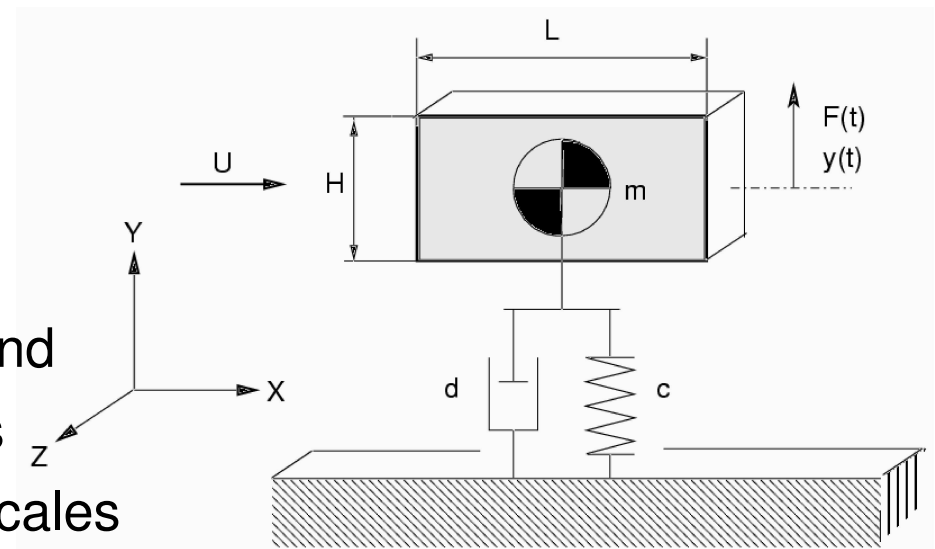




- Compute two dominant frequencies and corresponding amplitudes
 - Vortex shedding
 - Leading-edge vortices impinging on the trailing edge ($\frac{L}{H} \approx 2$)

- Unsteady RANS

- simple geometry, but complex flow and wide range of time and length scales
- overlapping modeled and resolved scales
- (standard) EVM usually too diffusive

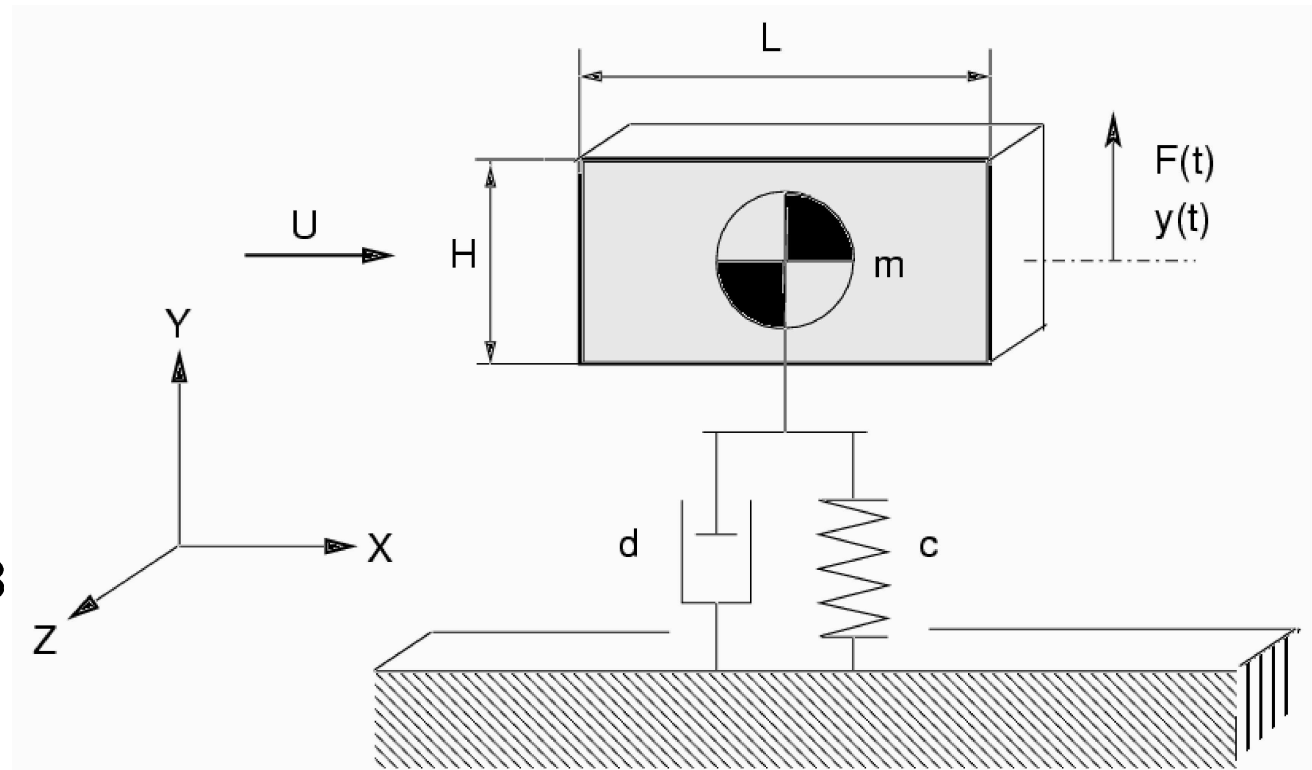


- Oscillation differential equation (1 dof)

$$m\ddot{y} + d\dot{y} + cy = F$$

- Testcase data

- $D = \frac{d}{4\pi m f_0} = 0.0032$
- $f_0 = \frac{1}{2\pi} \sqrt{\frac{c}{m}} = 3.27 \text{ Hz}$
- $L/H = 2$
- $Sc = 4Dm/(\rho_F HL) = 2.88$



- Discretization

$$y_{t+\Delta t} = f(y_{\Delta t}, y_{t-\Delta t}, y_{t-2\Delta t}, y_{t-3\Delta t}, F_t) + O(\Delta t^3)$$

- Analytical solution available for validation

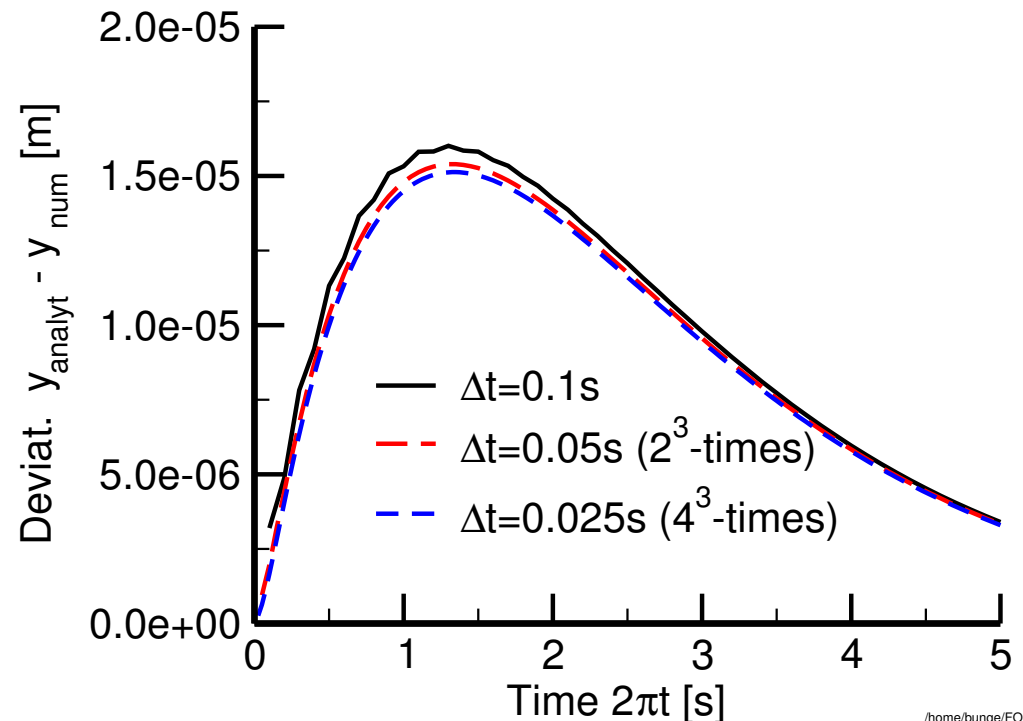
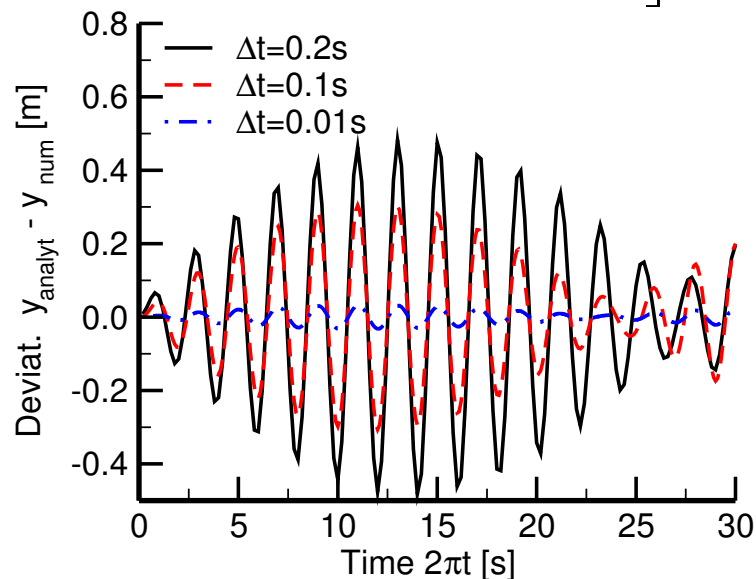


- Analytical solution available for validation

$$y(t) = Y[e^{-f_0 D t} \cos(f_D t + \varphi)] + A\left[\frac{1}{c} \alpha(\Omega) \cos(\Omega t - \delta(\Omega))\right]$$

- Discretized equation successfully validated and tested

$$y_{m+1} = \frac{-1}{\left(\frac{11m}{12(\Delta t)^2} + \frac{d}{3\Delta t}\right)} \left[y_m \left(\frac{-5m}{3(\Delta t)^2} + \frac{d}{2\Delta t} + c \right) + y_{m-1} \left(\frac{m}{2(\Delta t)^2} - \frac{d}{\Delta t} \right) + y_{m-2} \left(\frac{m}{3(\Delta t)^2} + \frac{d}{6\Delta t} \right) - y_{m-3} \left(\frac{m}{12(\Delta t)^2} \right) - F_m \right]$$



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STRUCTURE-MECHANICAL MODEL II

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

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- Reynolds averaged Navier Stokes equations (ALE)

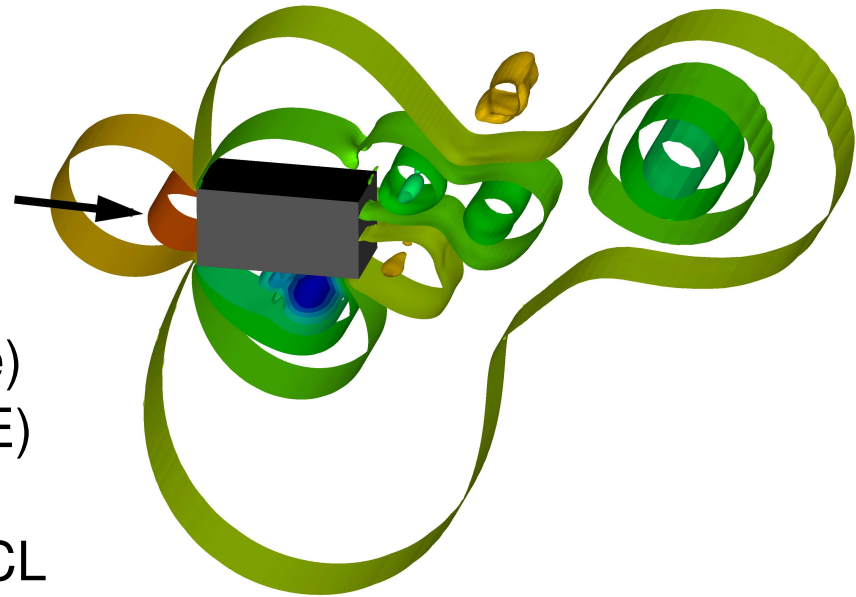
$$\int_V \frac{\partial U_i}{\partial t} dV + \oint_{\partial V} U_i (U_j - U_j^{\partial V}) dA_j = - \oint_{\partial V} \frac{P}{\rho} dA_i + \oint_{\partial V} \nu \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \overline{u'_i u'_j} dA_j$$

- Testcase data

- $Re = UH/\nu = 1 \cdot 10^4 \dots 6 \cdot 10^4$
- incompressible, fully turbulent

- Numerics

- Implicit FV method, cell-centered, semi block-structured (in-house code)
- Pressure correction method (SIMPLE)
- Conservative form in curvilinear, body-fitted coordinates employing SCL
- Second order in space (TVD) and time
- Variety of possibilities to treat turbulence



- Analytical solution not available



● Hybrider Ansatz aus RANS und LES

- Detached–Eddy Simulation (DES)
- Ersetzung des turbulenten Längenmaßes $l_{\text{turb}} = l_n$ oder $\frac{\sqrt{k}}{\omega}$ durch $\text{MIN}(l_{\text{turb}}; C_{\text{DES}}\Delta)$ im Dissipationsterm des Modells
- Kalibrierung der Konstanten C_{DES} mittels DIT im LES Modus
- Turbulenzmodell funktioniert wie SGS–Modell
- Keine zusätzlichen Übergangsbedingungen ('gray area')

● Hybride Numerik

- Hybrides Konvektionsschema (upwind basiert / zentral)
 - Kontinuierlicher Übergang durch Scherung und Rotation gesteuert
CDS in Bereichen grosser Wirbelstärke
UDS-basiert sonst (Wandnähe, Fernfeld)
- Numerische Stabilität in Bereichen elementarer Strömung
- Geringe numerische Diffusion in LES–Bereichen



Models rely on isotropic or linear eddy–viscosity principle

$$\overline{u'_i u'_j} = \frac{2}{3} k \delta_{ij} - \nu_t \underbrace{\left[\left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) - \frac{2}{3} \frac{\partial U_k}{\partial x_k} \delta_{ij} \right]}_{=2S_{ij}}$$

Spalart & Allmaras Model transport equation

$$\frac{\partial \tilde{\nu}}{\partial t} + U_i \frac{\partial \tilde{\nu}}{\partial x_i} = c_{b1} \tilde{S} \tilde{\nu} + \frac{3}{2} \left[\frac{\partial}{\partial x_i} \left((\nu + \tilde{\nu}) \frac{\partial \tilde{\nu}}{\partial x_i} \right) + c_{b2} \left(\frac{\partial \tilde{\nu}}{\partial x_i} \right)^2 \right] - (c_{w1} f_w) \left(\frac{\tilde{\nu}}{l} \right)^2$$

with

$$\tilde{S} = S \left(\frac{1}{X} + f_{v1} \right), \quad S = \sqrt{2S_{ij}S_{ij}}, \quad f_w = g \left[\frac{1 + c_{w3}^6}{g^6 + c_{w3}^6} \right]^{\frac{1}{6}}, \quad g = r + c_{w2}(r^6 - r),$$

$$c_{b1} = 0.1355, \quad c_{b2} = 0.622, \quad c_{w1} = \frac{c_{b1}}{\kappa^2} + \frac{(1 + c_{b2})}{\sigma}, \quad \kappa = 0.41,$$

$$r = \tanh\left(\frac{\tilde{\nu}}{\kappa^2 l^2 \tilde{S}}\right) / \tanh(1), \quad c_{w2} = 0.3, \quad c_{w3} = 2, \quad \sigma = \frac{2}{3}.$$

SALSA Model: modified production term, c_{b1} enhanced/dependent on strain rate, sensitized to non–equilibrium effects, reduction of production for excessive strain rates



Based on transport equation for turbulent kinetic energy k

$$\frac{\partial k}{\partial t} + \frac{\partial U_i k}{\partial x_i} = P_k - \beta_k \omega k + \frac{\partial}{\partial x_i} \left(\left(\nu + \frac{\nu_t}{2} \right) \frac{\partial k}{\partial x_i} \right)$$

and for turbulent frequency ω

$$\frac{\partial \omega}{\partial t} + \frac{\partial U_i \omega}{\partial x_i} = P_\omega - \beta_\omega \omega^2 + \frac{\partial}{\partial x_i} \left(\left(\nu + \frac{\nu_t}{2} \right) \frac{\partial \omega}{\partial x_i} \right).$$

Both models:

$$\nu_t = c_\mu \frac{k}{\omega}.$$

Models differ in constants β and representation of production terms

$$P_k = c_\mu \frac{k}{\omega} S^2 \quad \text{and} \quad P_\omega = \alpha S^2$$

LLR $k - \omega$ Model: c_μ , α determined in different ways, local linear,
derived from realizability and non-equilibrium constraints,

coefficients of stress–strain relation and transport equations are

functions of non-dimensional invariants of mean strain and vorticity rates.

(approach tries to accomplish consistent stress–strain distributions not only in
plane shear flow, but also in more general flow situations)

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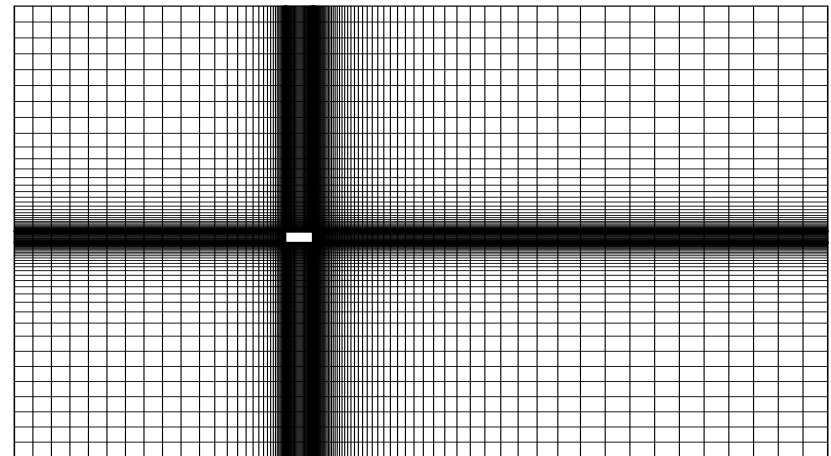
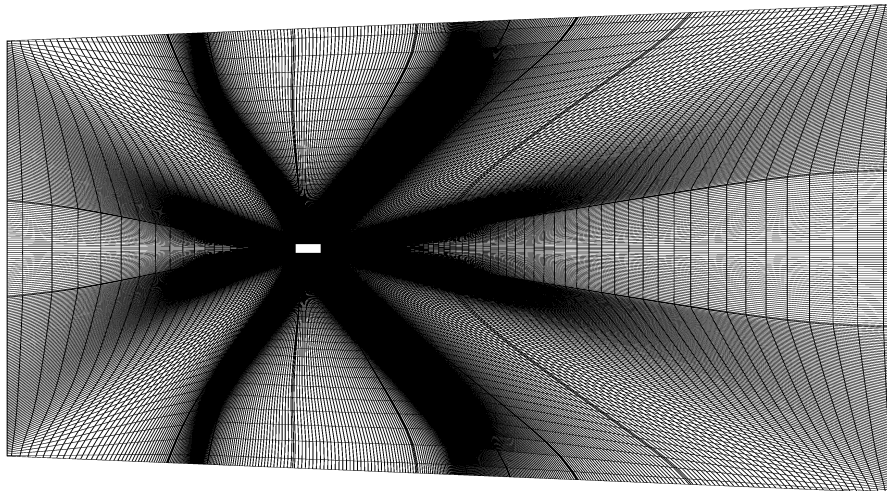
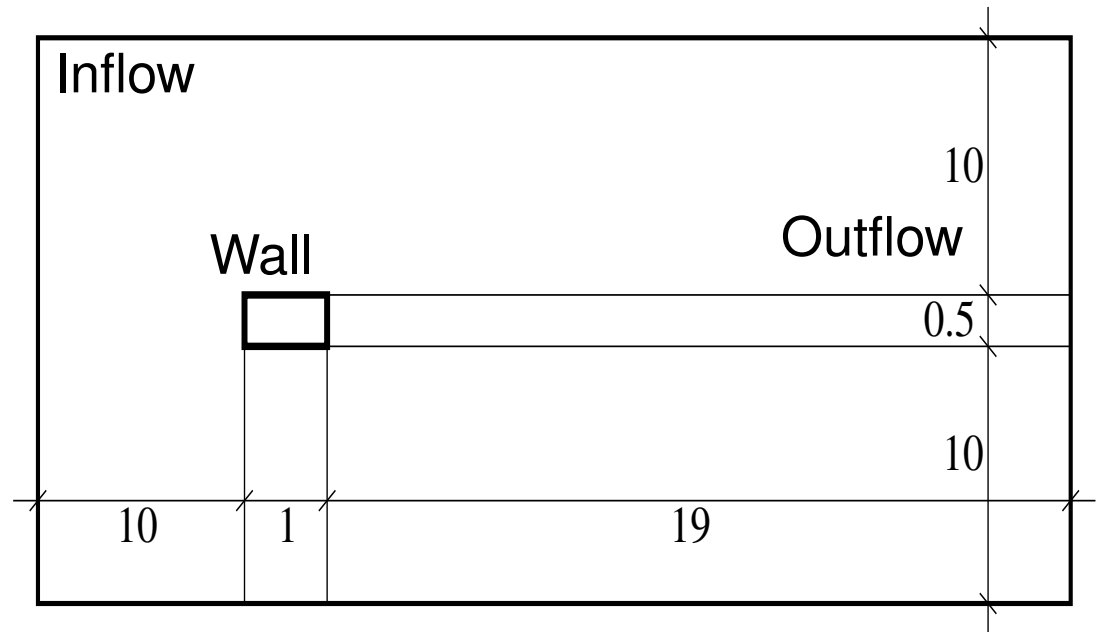
TWO-EQUATION MODELS

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

U. Bunge, A. Gurr, F. Thiele



- Different 2D grids
20,000 to 80,000 volumes
rectangular, curved, ...
- Different 3D grids, $\Delta z = 1, 2$
extended 2D grid, 10-40 levels
up to $\sim 9,000,000$ volumes
- $Y^+(i_n = 1) = l_n \sqrt{\frac{1}{\nu} \frac{\partial u_t}{\partial Y_n}} \leq 1$



NUMERICAL GRIDS FOR FLOW SOLUTION

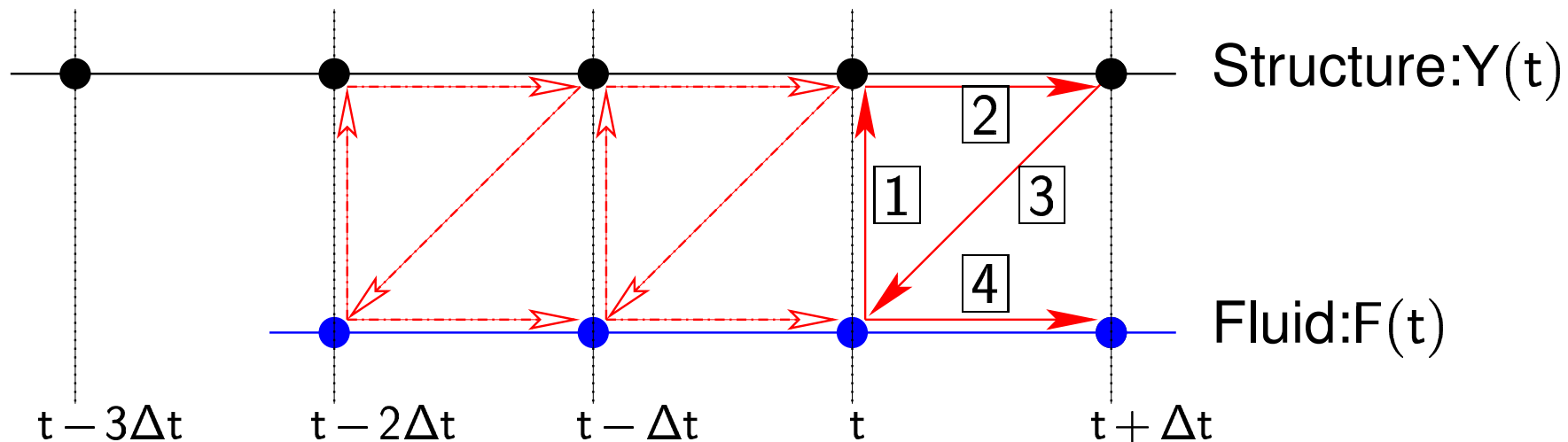
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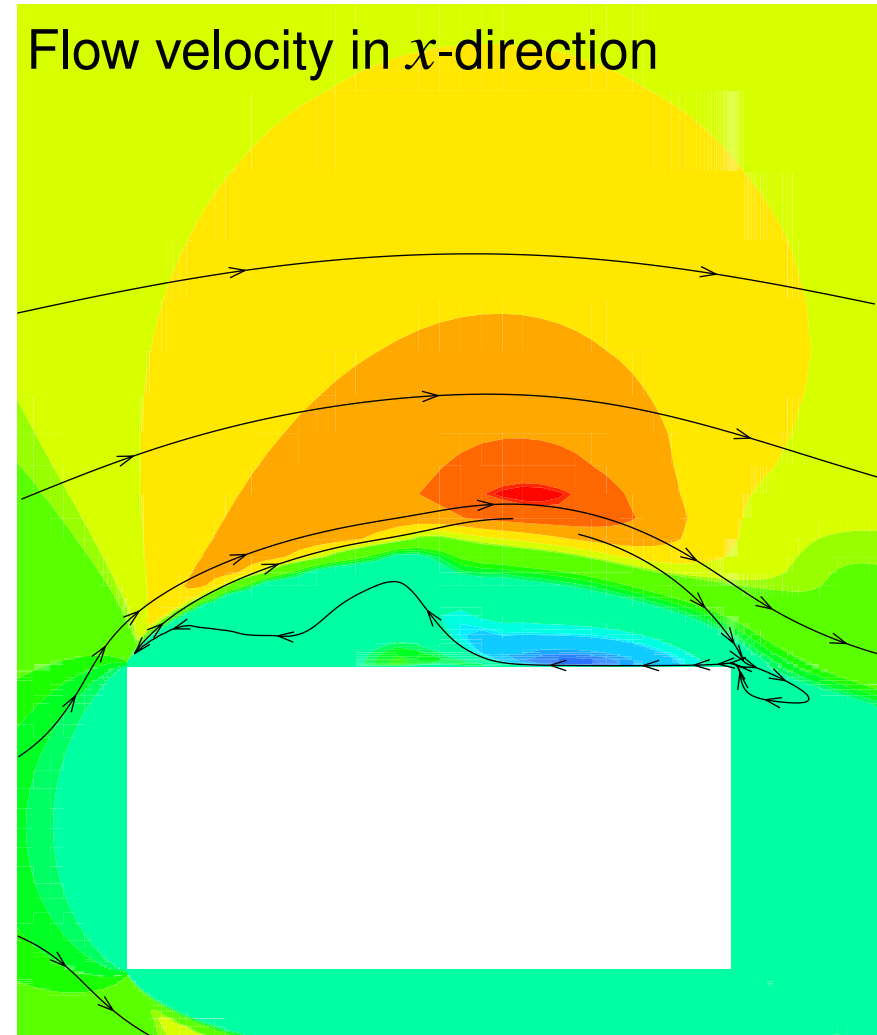
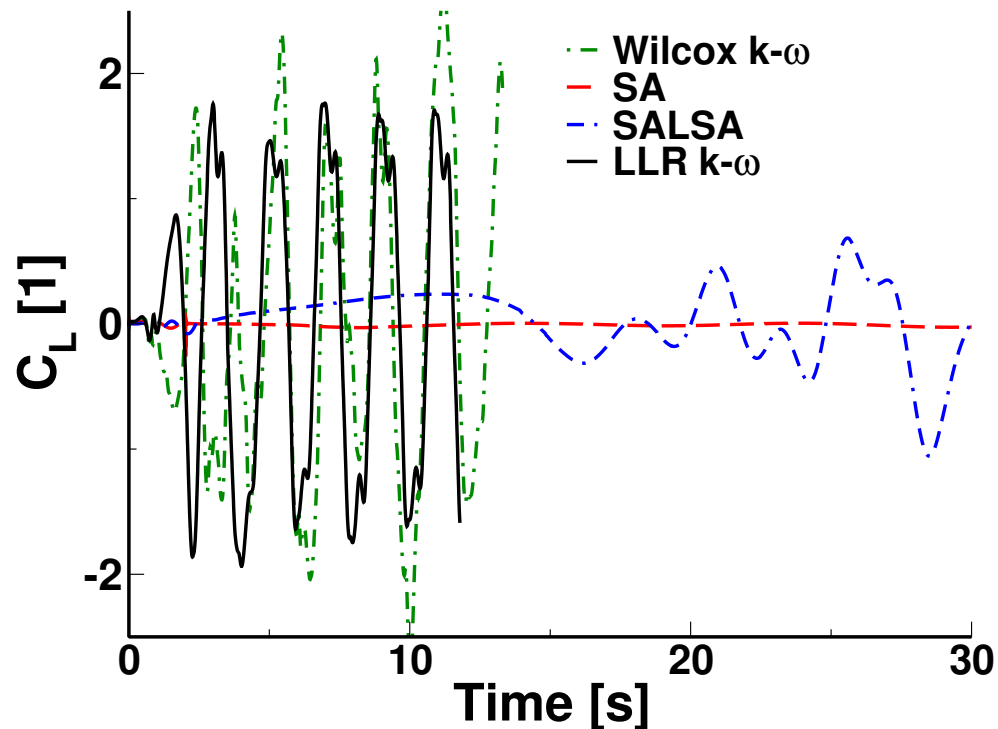
- Explicit, loose coupling
- First order in time
although single procedures are of higher order
- Sufficient in this case
small time step required for flow solution

$$T_s/\Delta t \geq 200$$



- Turbulence models

- SA and SALSA
- Wilcox $k-\omega$ and LLR $k-\omega$

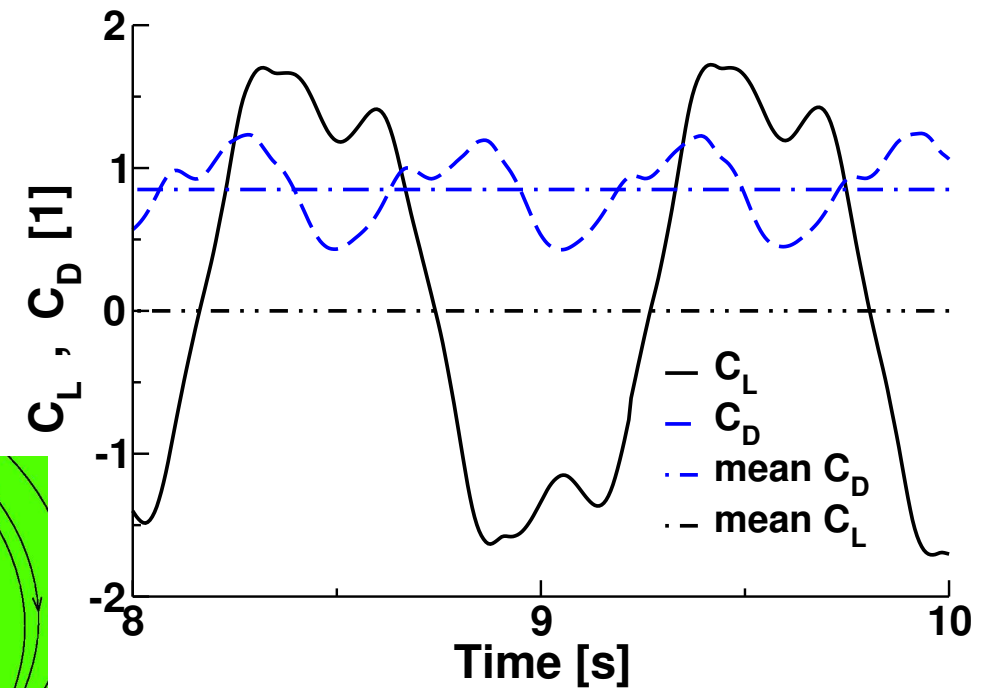
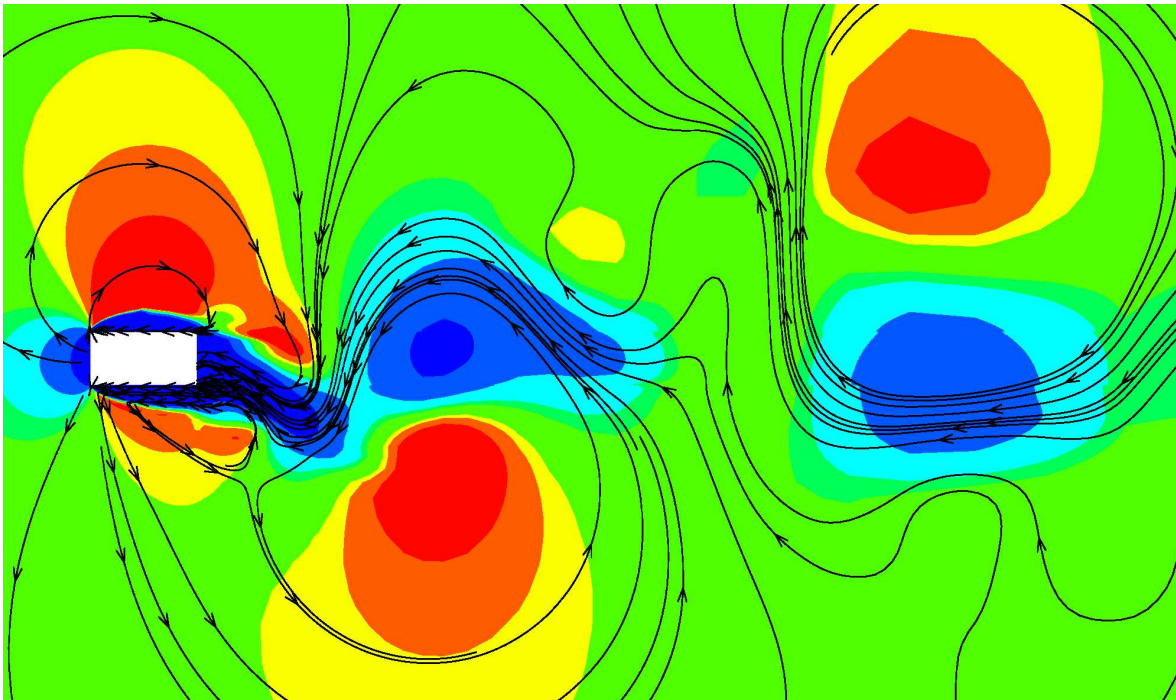


RESULTS – STATIONARY BODY 2D / 3D

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

U. Bunge, A. Gurr, F. Thiele

- Vortex shedding occurs
- Mean lift vanishes
- Agreement with experiments (except for frequencies)



- Mean $C_D \approx 0.85$
- Max drag = Max lift
- Min drag = dent in lift due to recirculation

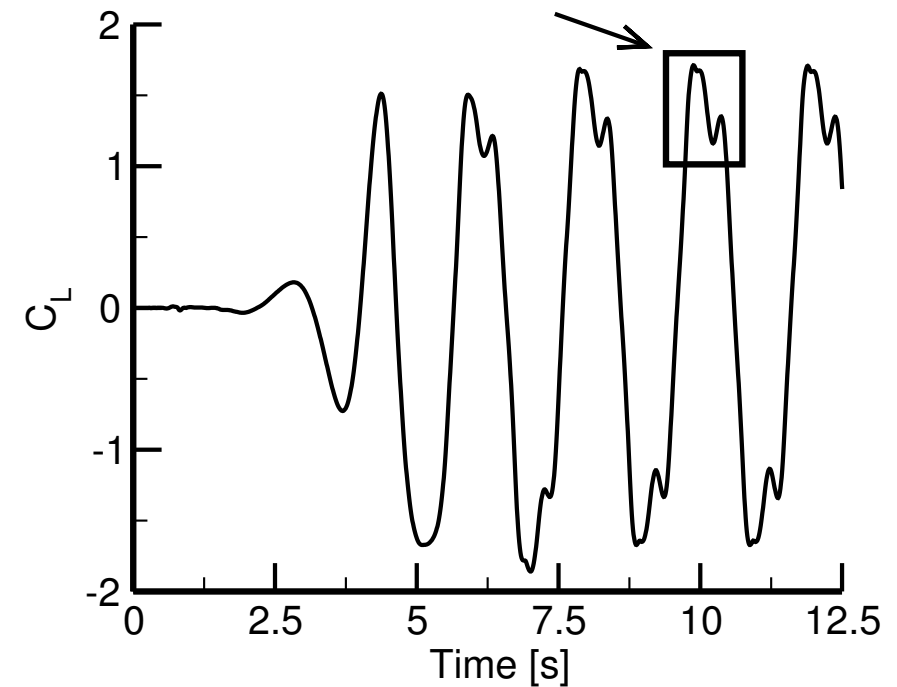
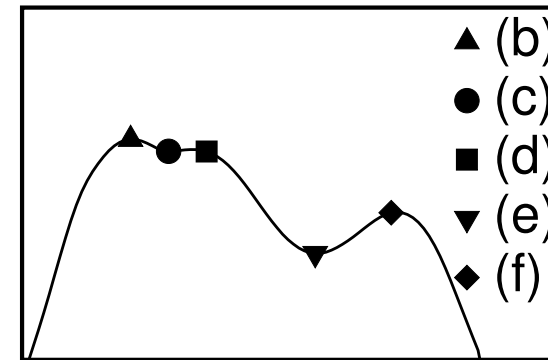
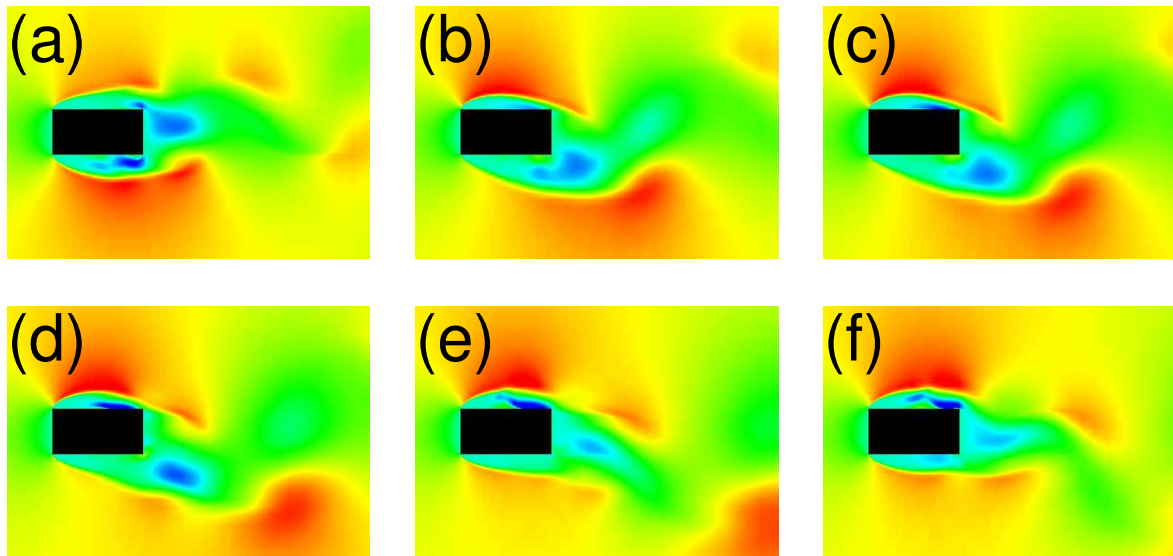


RESULTS – STATIONARY BODY 2D / 3D II

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

U. Bunge, A. Gurr, F. Thiele

- (a) close to zero lift
- (b) maximum lift
- (c) - (e) trailing edge interference
- (f) disturbed vortices



RESULTS – STATIONARY BODY 2D / 3D III

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

U. Bunge, A. Gurr, F. Thiele



- Resonance possible

- $f_0 = f_s \quad V = \frac{Uf_0}{H} = \frac{1}{St_s}$
- $f_0 = f_l \quad V = \frac{1}{St_l} < \frac{1}{St_s}$

- Experimental results

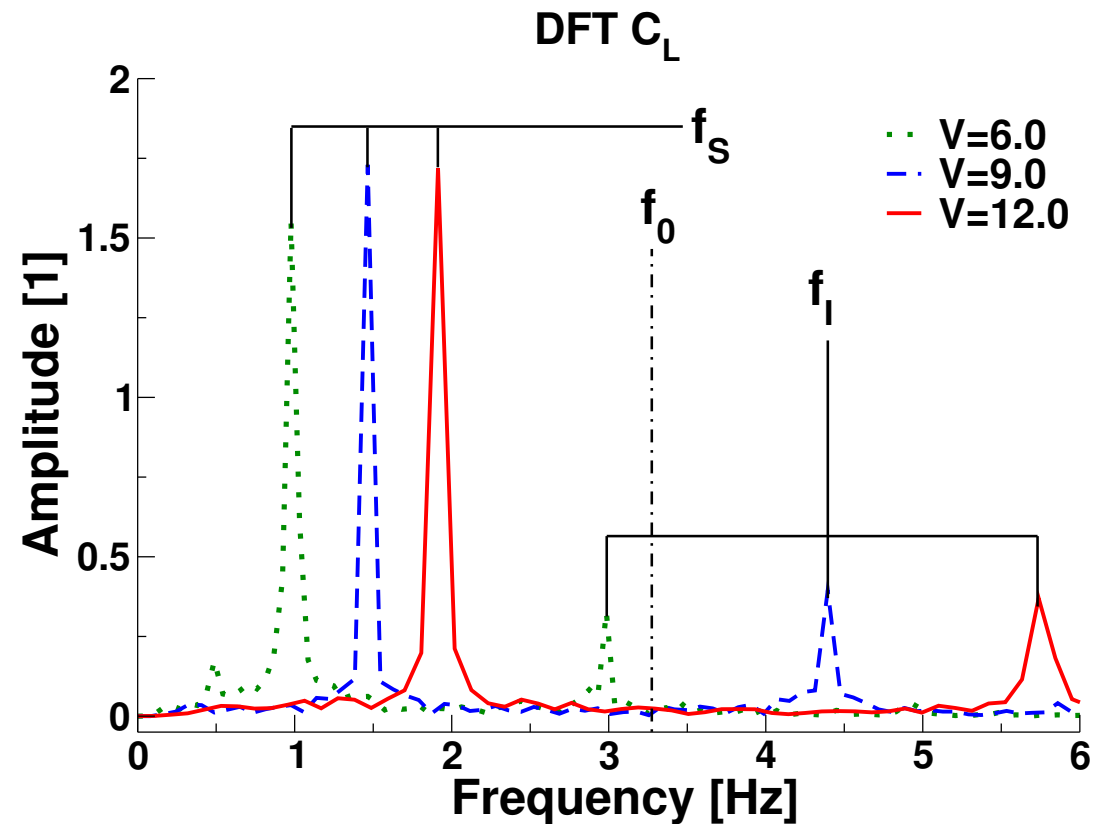
- $St_s \simeq 0.08$
- $V = 5.5$ first resonance

- Numerical results

- $St_s \simeq 0.06$
- Variation of
grid: 2D 80,000 CV

time step: $0.015s (V=3.0) > \Delta t_{dim} > 0.0015s (V=18.0)$

turbulence model: EVM only, one- and two-equation models



RESULTS – STATIONARY BODY 2D / 3D IV

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

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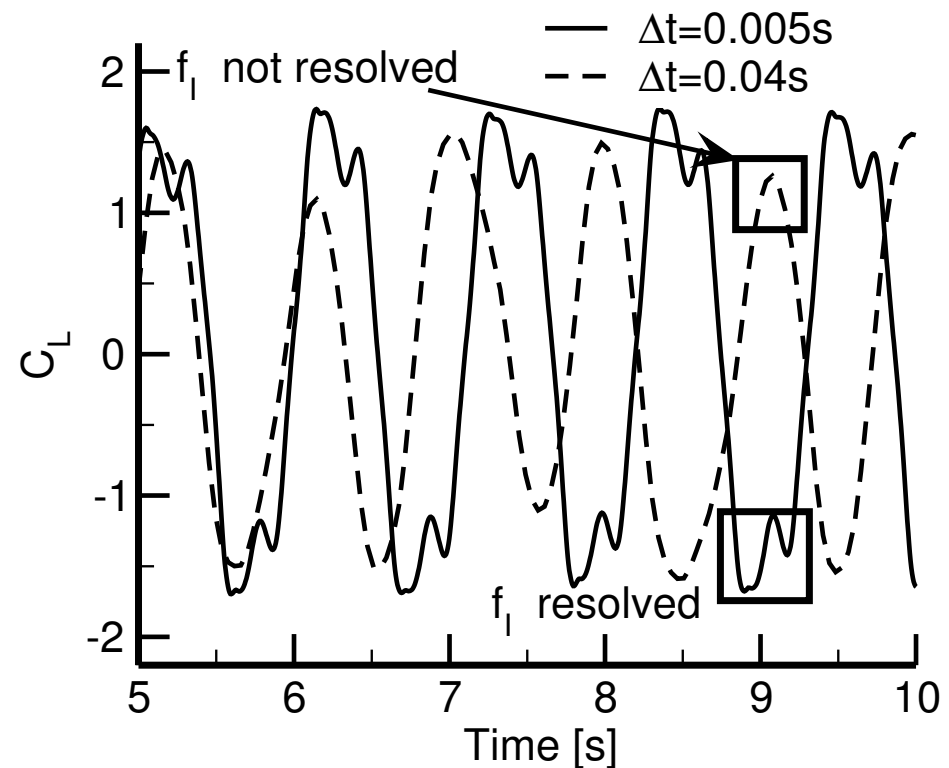
Parameters evaluated

● Numerics

- Temporal resolution
(Conclusion: $\Delta t \frac{U}{H} = \text{const}$)
- Spatial resolution
- Convection scheme
- Convergence criteria
- Relaxation
- Coupling of equations
- ...

● Physics

- Treatment of turbulence
- Boundary conditions



$V = \frac{U}{Hf_0} =$	3.0		5.5		15.0	
$\Delta t_{\text{dim}} [\text{s}]$	$\Delta t \frac{U}{H}$	s	$\Delta t \frac{U}{H}$	s	$\Delta t \frac{U}{H}$	s
0.005	0.05	0.055	0.092	0.053	0.25	0.057
0.01	0.1	0.055	0.183	0.050	0.5	0.055
0.02	0.2	0.057	0.366	0.061	1.0	0.054
0.04	0.4	0.062	0.732	0.067	2.0	0.033

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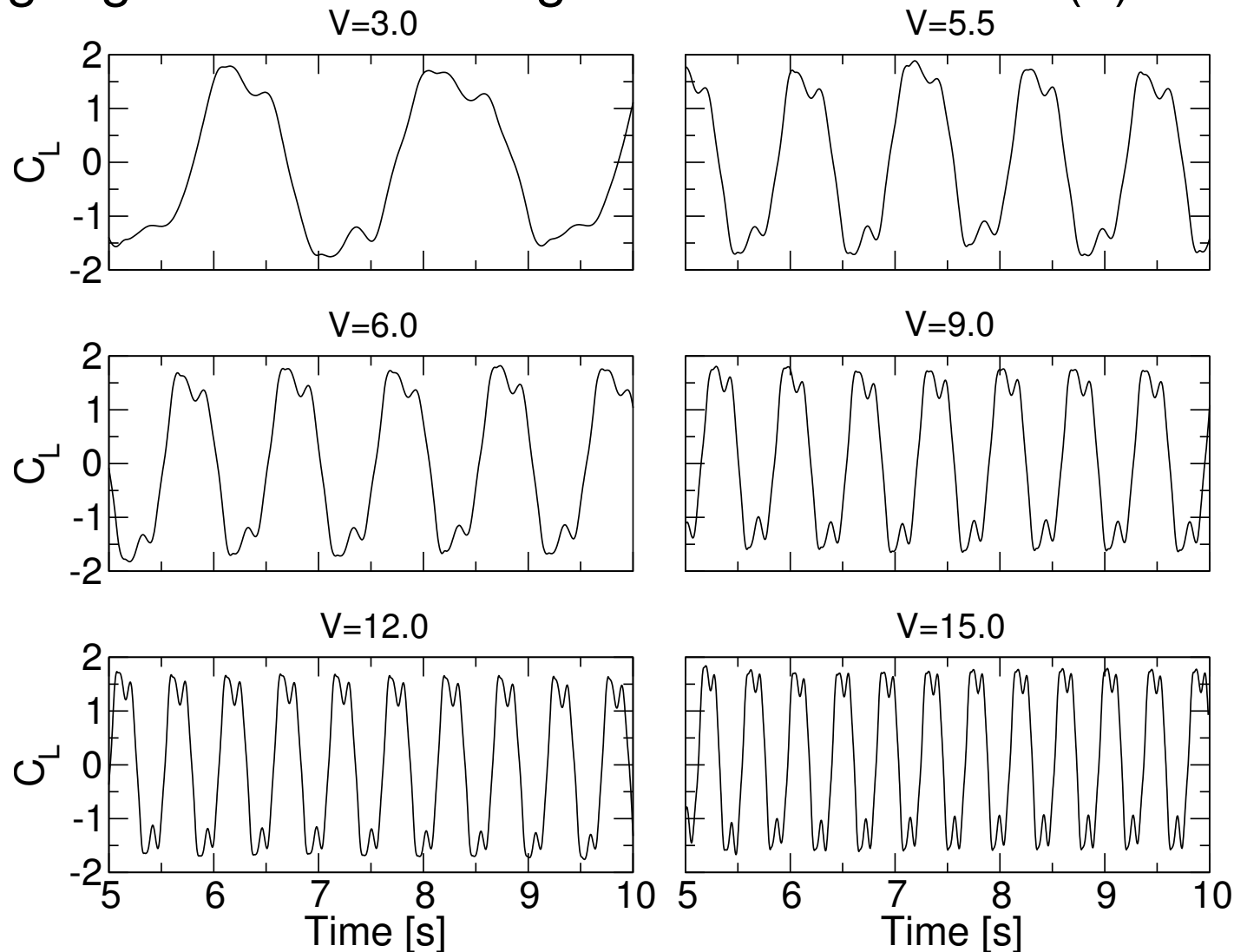
RESULTS – STATIONARY BODY 2D / 3D V

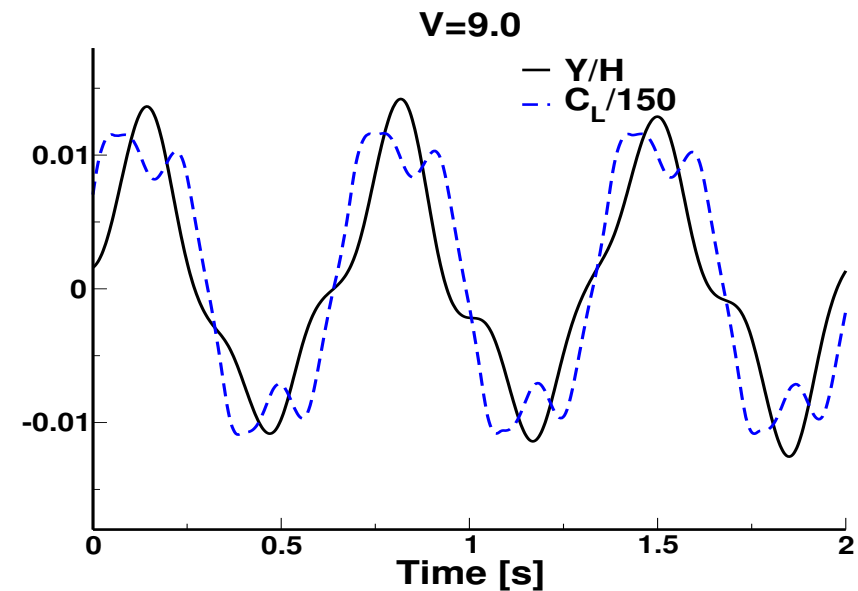
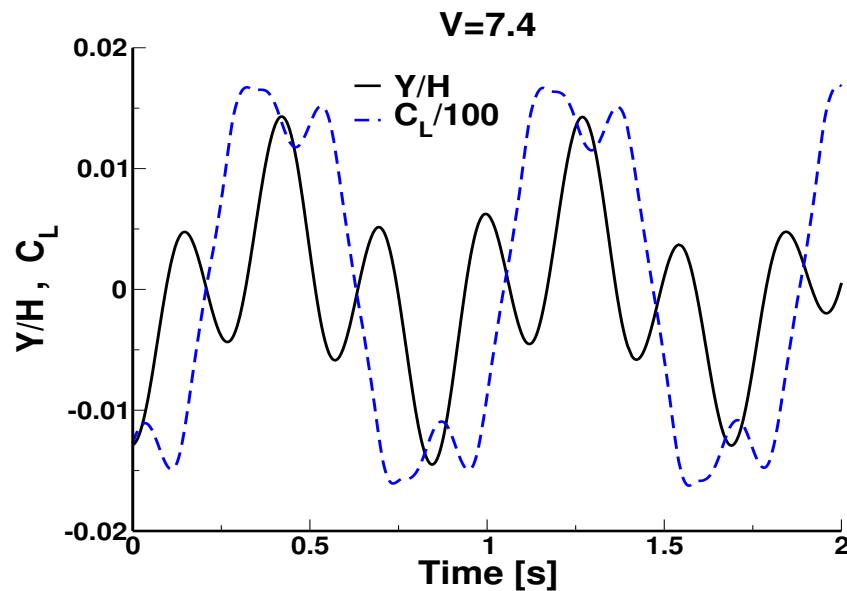
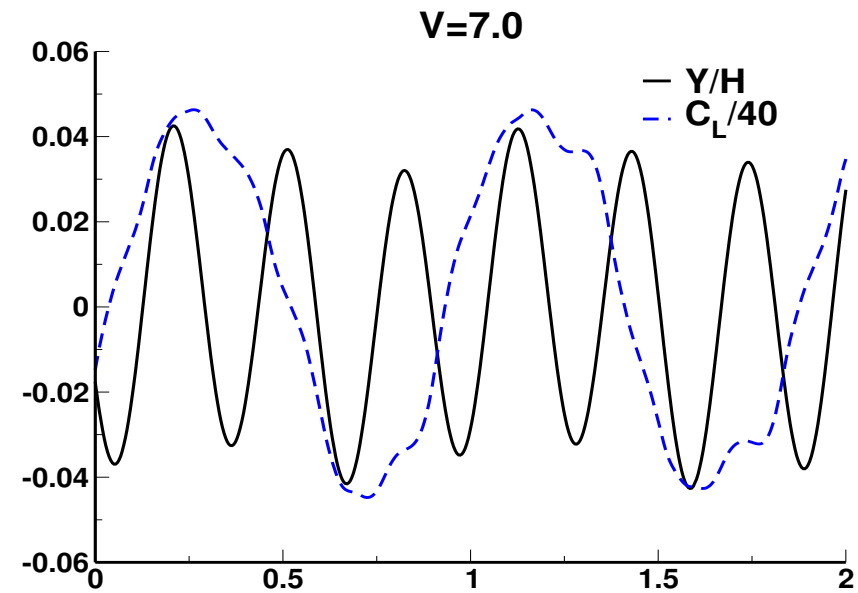
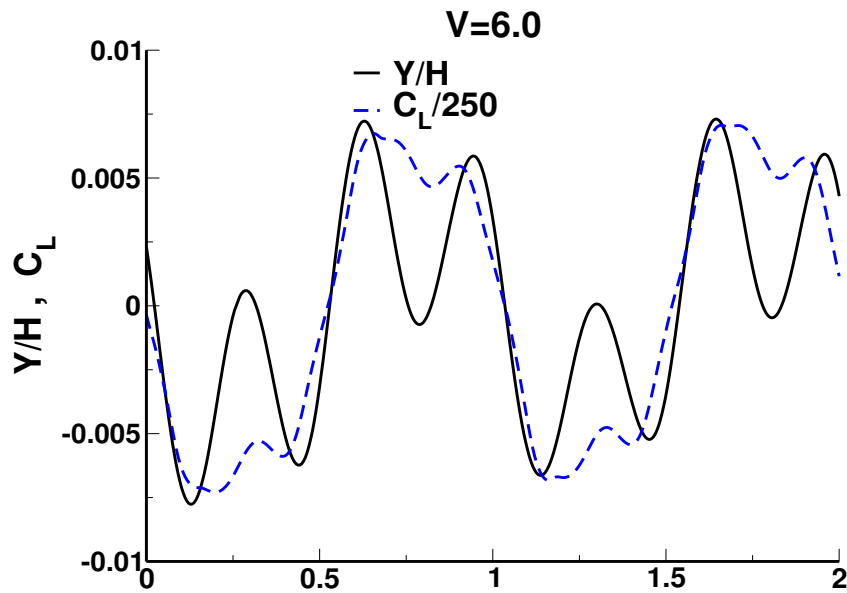
FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

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- Bewegung: Keine Änderung im Auftriebsverlauf (?)





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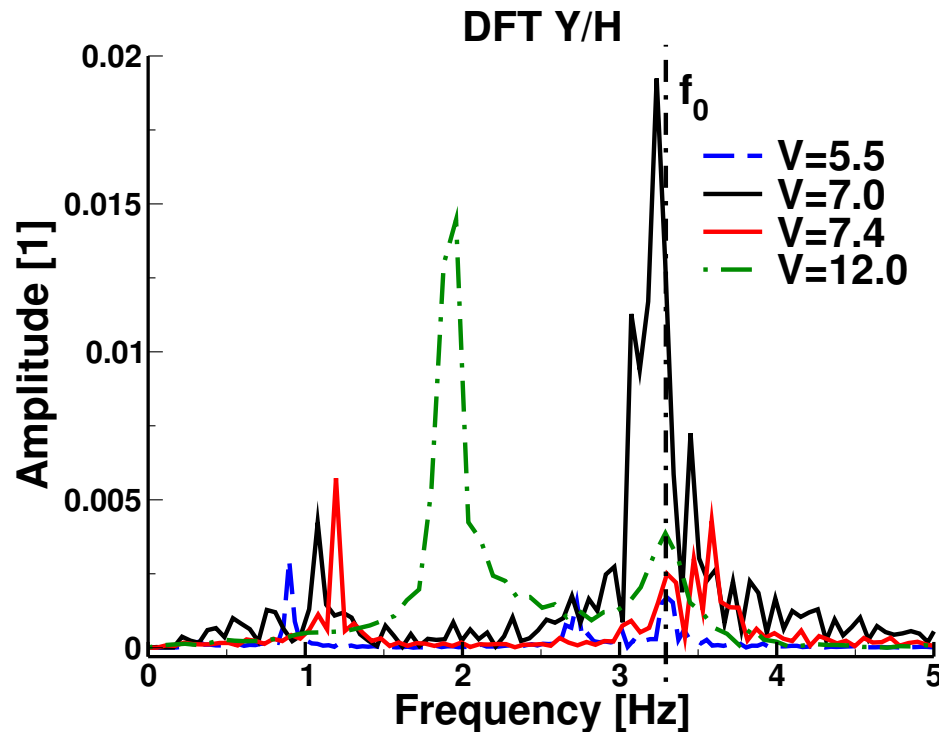
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RESULTS – MOVING BODY 2D II

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

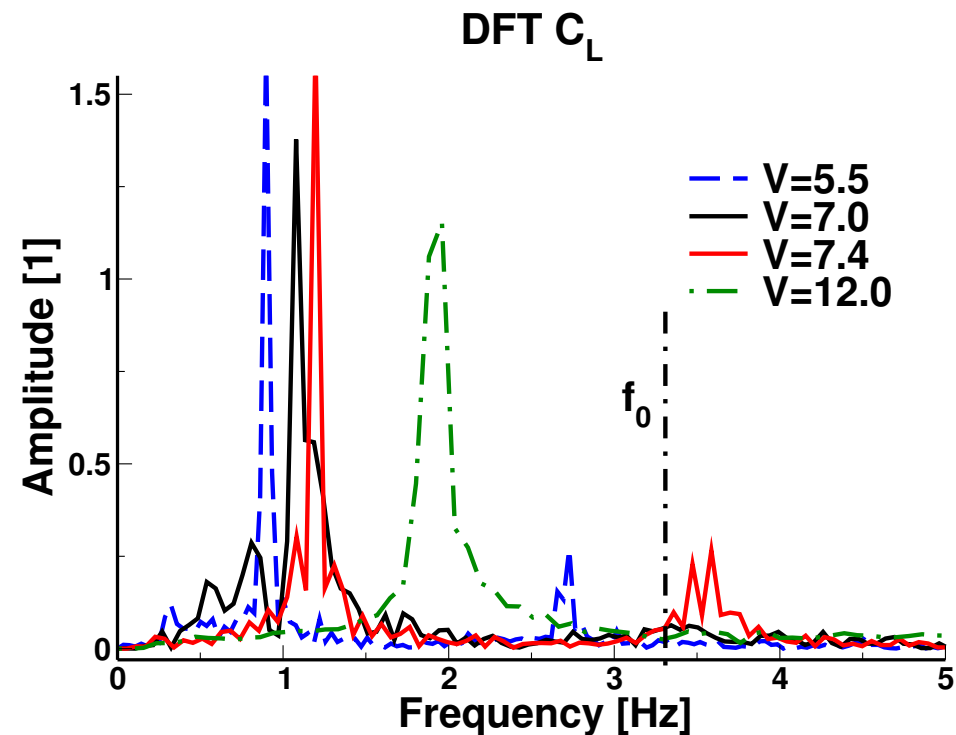
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⇒ Excitation reveals peak value
Resonance flow velocity:
 $V \approx 7.0$

Lift peak value vanishes ⇐
Interaction obvious
Frequency predictable
without movement



- Resonance velocity

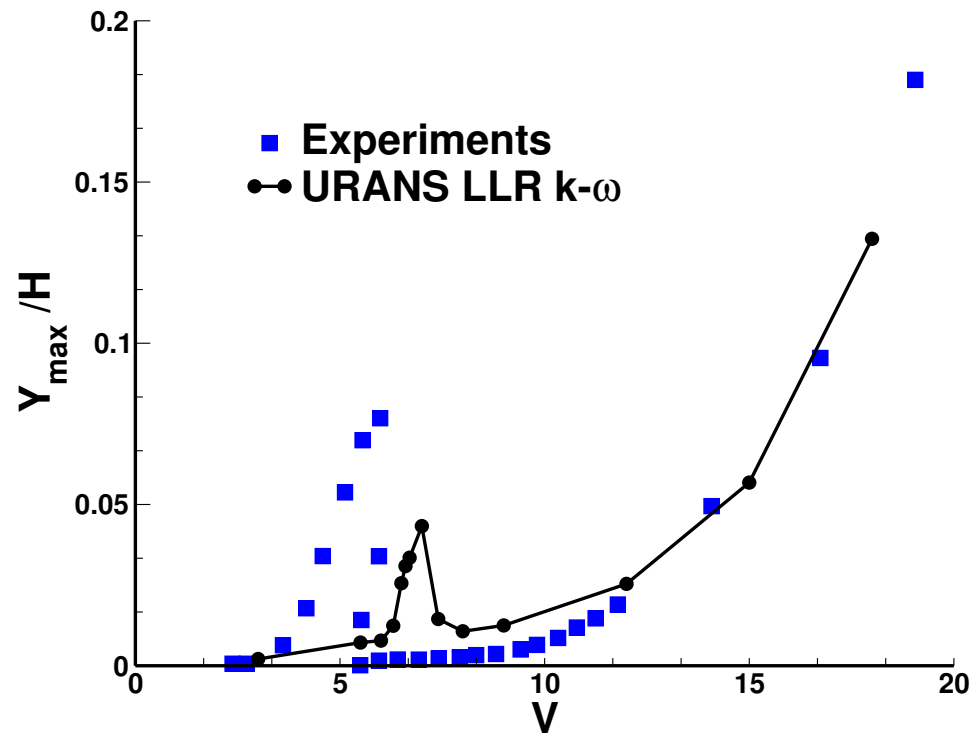
- $f_{\text{Inum}} < f_{\text{Iexp}}$
- Too much damping (viscosity)
- Principal problem of URANS

- Maximum excitation

- Also too much damping
- Coupling procedure

- Experimental results

- Confined to one case
- Missing information on turbulence intensity
and wind tunnel geometry



- Galloping velocity is captured

- $V \approx 12.0$

- Criterion:

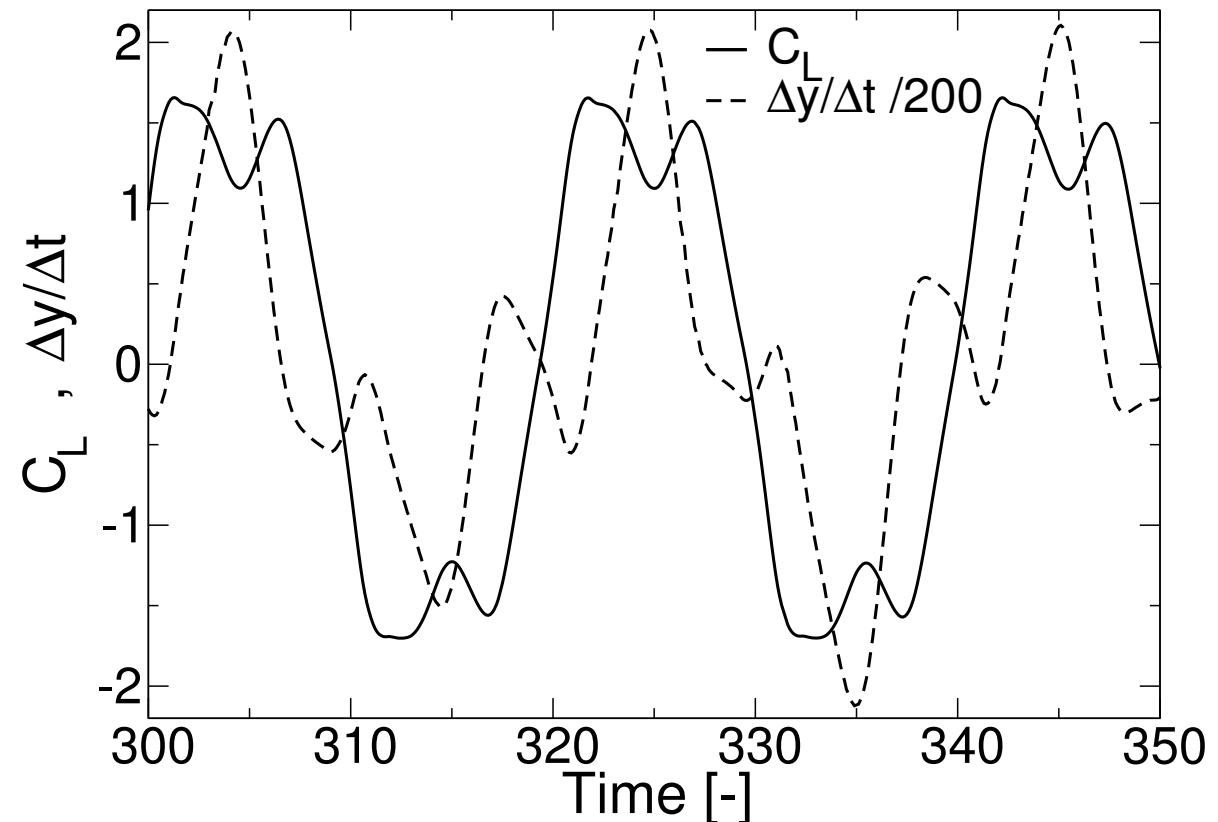
$$\frac{V}{4\pi Sc} + \frac{2}{\frac{\partial C_L}{\partial \alpha} \cos(2\pi \frac{\tau_r}{V})} = 0$$

- Scruton number needed

- $\frac{\partial C_L}{\partial \alpha}$ from experiments

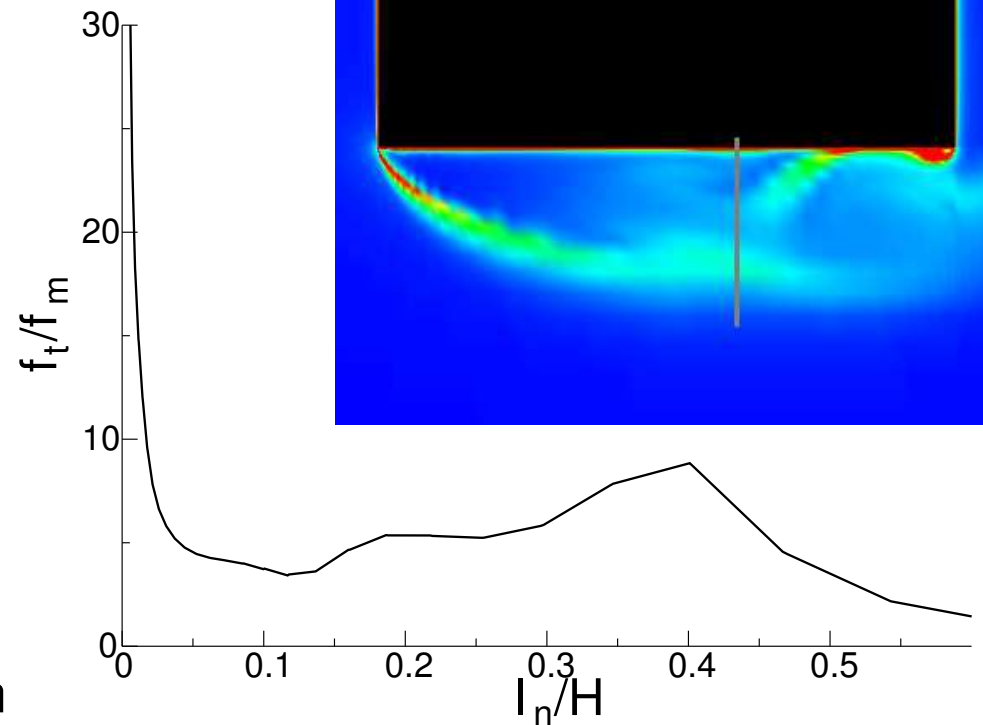
- τ_r = Time lag between lift and $\frac{\partial y}{\partial t}$ estimated \rightarrow

- However, frequency to low



● Comparison of f_t and f_m

- Turbulent frequency
 $f_t \sim \omega$, turbulent time scale
- Maximum resolved frequency
 $f_m \sim 1/\Delta t$, time step
- Gap exists up to
10 CV from the wall
- Minimum f_t in the wake
7...15 H from the wall
- Overlapping
in the range of $O(10^3)$
even close to the wall
- Hybrid methods should be chosen
despite increased computational costs (DES)



RESULTS – SPECTRAL GAP PROBLEM

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

U. Bunge, A. Gurr, F. Thiele

- Physical phenomenon captured very well by 2D URANS
 - Validity of URANS doubtful when spectral gap not adhered to
- Frequencies are too low even for uncoupled (stationary body) computations
 - Coupling procedure sufficient
 - Drawback of methods (usually) add up (small Strouhal number)
 - Flow physics modeling has to be more sophisticated
 - Simple (one-equation) and unmodified standard two-equation turbulence models perform even worse
- Fine grids and schemes of higher order necessary
- 2D and 3D computations give equal results
 - So far only the same methods and models have been tested
- Not enough experimental data available for this case
- More effects have to be examined (to confirm results)



CONCLUSIONS

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

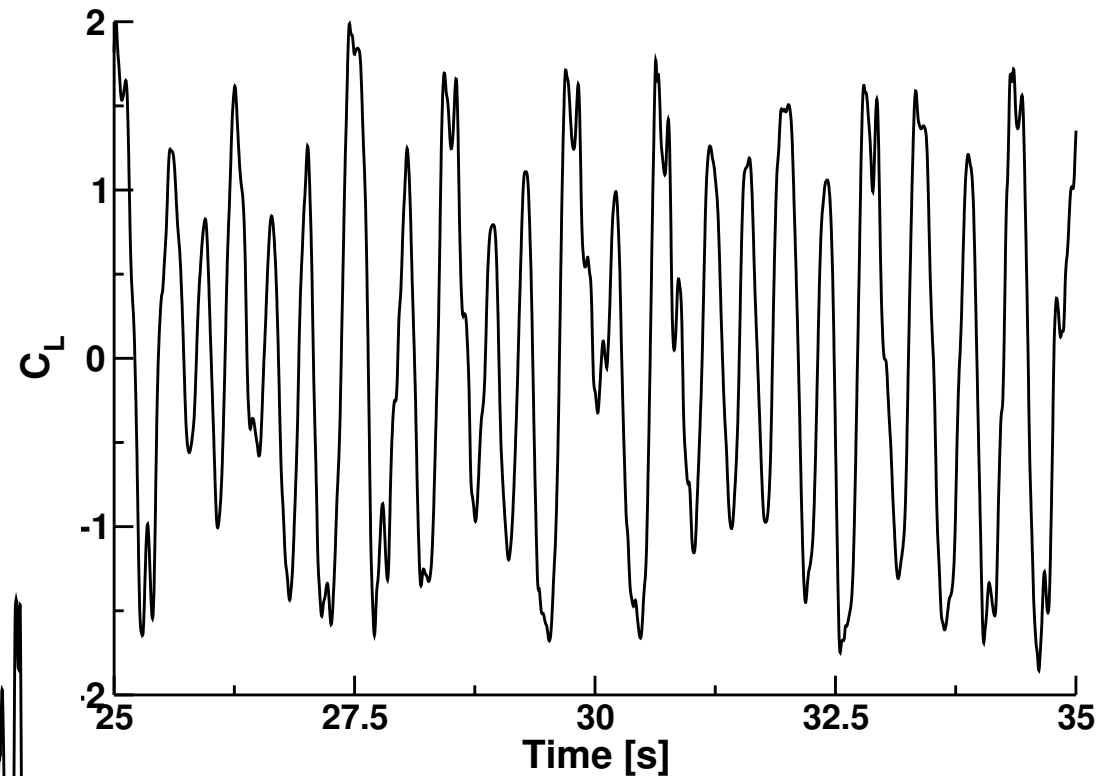
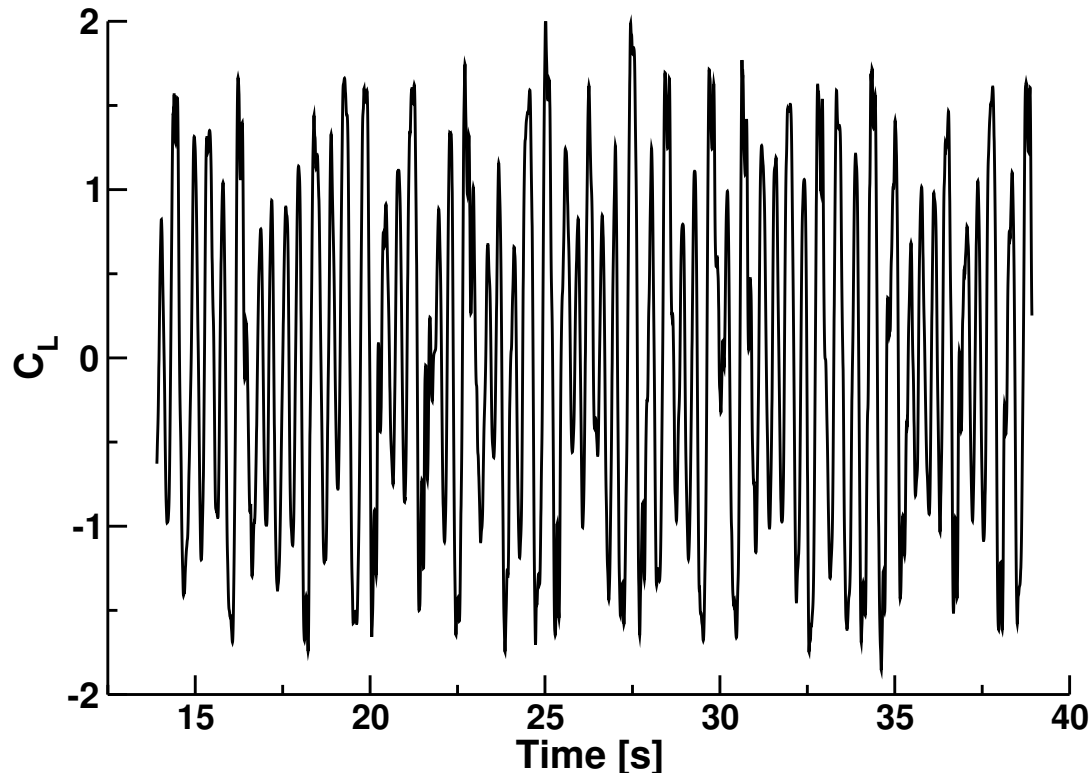
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- More parameters will be examined:
 - Geometric modeling, i.e. wind tunnel walls in different positions
 - ⇒ Movement of body will have more influence
 - ⇒ Deforming grids are necessary
 - Flow physical modeling
 - ⇒ 2D URANS Limitation of v_t
 - ⇒ Sophisticated models (EARSM), curvature correction
 - ⇒ 3D DES using EVM + EARSM
- Experimental data: any hint or collaboration is welcome
- Areas of application for URANS in FSI limited ! :
 - Aerodynamic of buildings (bridges)
 - Airfoil flutter (compressible)
 - Unsteady flows in general



- Neue Ergebnisse für DES
 $V=5.5$, erste Auswertung
 2,000,000 nur ausgewertet
 $\Delta z = 1$ mit 15 Ebenen



- Verhalten unregelmäßiger
- Amplituden ähnlich
- 3D Effekte schon sichtbar trotz groben Gitters (ohne Bild)



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NEW RESULTS – STATIONARY BODY 3D (DES)

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

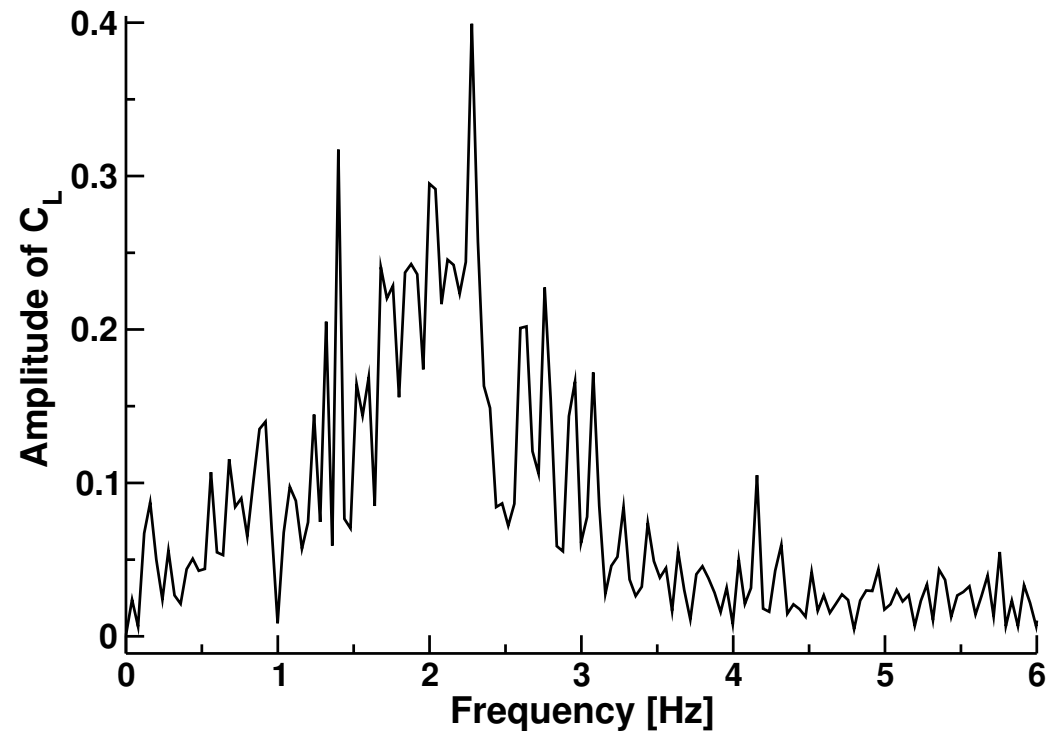
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- Frequenzanalyse

- Frequenzen höher,
nahe am Experiment
- Breiteres Spektrum,
keine ausgezeichneten peaks
- Resonanzverhalten schwerer vorhersagbar
- Kopplung mit schwingungsfähigem Körper als Filter (?)



NEW RESULTS – STATIONARY BODY 3D (DES) II

FLOW-INDUCED OSCILLATIONS OF A BLUFF BODY

U. Bunge, A. Gurr, F. Thiele

